Constellation

SPIE 5168-35

Constellation-X Spectroscopy X-Ray Telescope Optical Assembly Pathfinder Image Error Budget and Performance

Prediction

Dr. William A. Podgorski(SAO)

Co-authors:

Dr. Jay Bookbinder (SAO)

Dr. David Content (NASA/GSFC)

Mr. William Davis (SAO)

Mr. Mark Freeman (SAO)

Mr. Jason Hair (NASA/GSFC)

Dr. Scott Owens (NASA/GSFC)

Dr. Robert Petre (NASA/GSFC)

Dr. Paul Reid (SAO)

Dr. Timo Saha (NASA/GSFC)

Mr. Jeffrey Stewart (NASA/GSFC)

Dr. William Zhang (NASA/GSFC)

August 7, 2003

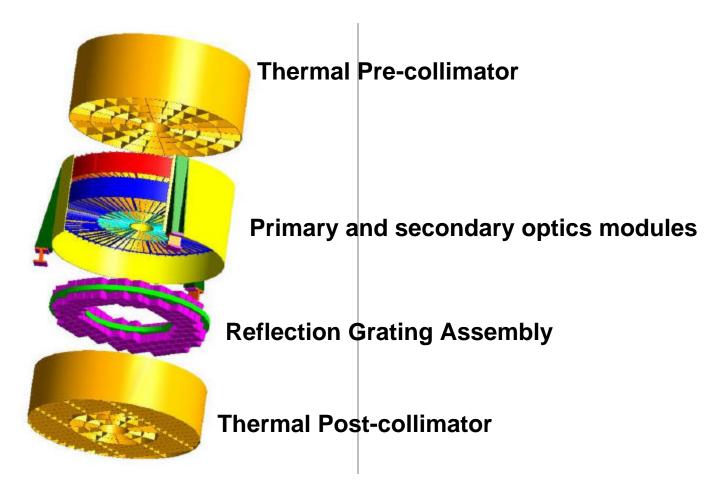


Overview

- SXT Flight Mirror Assembly (FMA) has top level requirement for Image Resolution of 15 arc-seconds (HPD)
 - FMA Error Budget Includes Contributors for:
 - As-built FMA (~12 arc-seconds HPD)
 - FMA on-orbit errors
 - Telescope errors
 - Detector errors
 - Co-add effects (4 satellites)
- OAP2 is second in a series of developmental SXT optical pathfinders (see 5168-21, R. Petre et. Al.):
 - Specified to have the same <u>as-built</u> image resolution as FMA but for a single shell
 - Plan to evaluate image resolution of OAP2 in x-ray test
 - Gravity and thermal effects in the x-ray test could limit our ability to demonstrate image resolution performance
 - Will report on work done to evaluate these effects.



Overview of SXT Flight Mirror Assembly (FMA)





SXT Flight Mirror Assembly - Image Error Budget

Contributors (HPD - arcsec)	Ramt	Margin	Allocations		Rationale				
RGS Resolution	15.00	3.92	14.48					4 satellites, post-processed	
Co-add 4 satellites				1.00				Superposition of data using X-ray centroids	
On-Orbit Telescope - single satellite				14.12				RSS	
CCD pixelization error				0.41				0.5 arcsec pixels	
Grating resolution error				3.00				Estimate	
XMS Resolution	15.00	4.95	14.16					4 satellites, post-processed	
Co-add 4 satellites				1.00				Superposition of data using X-ray centroids	
On-Orbit Telescope - single satellite				14.12				RSS	
Calorimeter pixelization error					4.08			5 arcsec pixels	
Telescope level effects					5.20			RSS	
- Image reconstruction errors (over obs)						4.24		RSS	
- SXT/Telescope mounting strain						2.00		Eng. estimate based on Chandra experience	
- SXT/SI vibration effects						2.00		Chandra experience (iitter)	
- SXT/SI misalign ment (off-axis error)						1.00		Chandra experience	
- SXT/SI focus error						0.20		Analysis	
SXT FMA - on-orbit performance					12.48			RSS	
- SXT FMA launch shifts						2.00		Eng. est. based on Chandra	
- Thermal errors						2.24		RSS	
- Material stability effects						1.00		Est. based on Chandra work	
- SXT FMA, as built						12.07		RSS	
Gravity release							1.50	FEA analysis using vertical assy	
Bonding strain							3.00	Eng. estimate, analysis in process	
Alignment errors (using CDA)							3.38	RSS	
Installation in housing							5.00	Est. based on OAP1 testing	
Optical elements							9.90	Est, based on tech dev program	

Requirement – 15 arc-seconds HPD

Constellation-X



Budget for Optical Elements

Optical Elements - Budget Allocations								
Error Term	Allocation	Sensitivity	Allocation	Units	Definition			
	HPD	HPD/Unit	Unit	12				
Optical Elements-Total	9.90							
Sag	3.00	25.00	0.12	μm	Peak-Valley			
Axial Figure	9.07	3.84	2.36	arc-sec	RMS Slope errors			
Roundness	0.75	0.15	5.00	μm	Peak-Valley			
ΔΔR	2.47	3.50	0.71	arc-sec	Peak-Valley			
Circularity	0.28	0.03	9.90 µm		See 2.2.4			
Average Radius	0.00	0.00	±33.00	μm	See 2.2.5			
Cone Angle	0.00	0.00	±10.00	arc-sec	See 2.2.5			

Table 1 - Optics Error Allocations

Largest contributor is Axial Figure

-Constellation-X



SXT OAP2 Image Error Prediction

Contributors (HPD - arcsec)		Est	imate/Pred	diction	Rationale/Comments	
Image Resolution - OAP2 X-Ray Test	14.34		1		RSS	
X-Ray Test Errors	Sc.	7.42			RSS	
Gravity Distortion	Į.	10	7.00			
Test Equipment errors			1.00			
Finite Source Distance	8	13	1.00	- P	ř .	
X-ray Source spot size		6 r	2.00		1 mm spot	
OAP2 Thermally Driven Errors		1.48	8		RSS	
Average Temperature Offset	Û		0.30		1 Degree C Temperature offset	
Epoxy/glass bi-layer effects	Ü		1.00		20µm EP301/400µm glass, 1 Degree C Offse	
Axial Gradient			0.10		0.5 degree end-to-end gradient	
Transverse Gradient			0.73		0.1 degree inner wall to outer wall gradient	
Vertical Gradient			0.75		0.5 degree top to bottom gradient	
Material Stability effects		1.00			Estimate	
OAP2 As Built		12.07			Analysis	
Gravity release	2	53	1.50	*	FEA analysis using vertical assy	
Bonding strain		9	3.00		Eng. estimate, analysis in process	
Alignment errors (using CDA)	88	4 ·	3.38		RSS	
CDA Dynamic Accuracy	i i		0	0.76		
CDA Static Accuracy	Ü		Sh CV	1.68	0	
Thermal Drift				2.00		
Adjustment Accuracy				2.00		
Installation in housing	Sc.	116	5.00		Est. based on OAP1 testing	
Optical elements			9.90		Est, based on tech dev program	

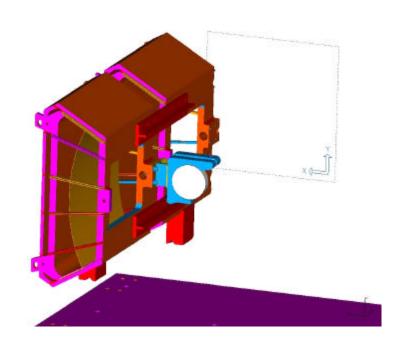


SXT OAP2 Image Error Contributors

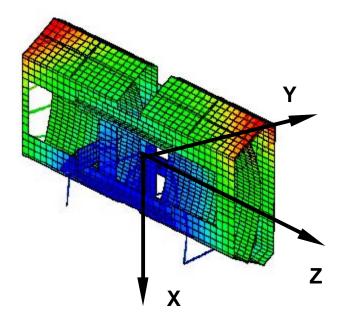
- Error contributors unique to testing OAP2 on the ground:
 - X-ray test errors including large gravity error
 - Thermally driven errors different from on-orbit thermal
- Analysis work performed to select best x-ray test orientation and evaluate impacts of 1G on image resolution
- Analysis work done to evaluate OAP2 image resolution sensitivities to thermal effects:
 - Average temperature offset from assembly temperature
 - Temperature gradients
- Thermal control system needed to limit these thermal impacts
- Planning done to evaluate these effects during x-ray testing



OAP2 and OAP2 Finite Element Model



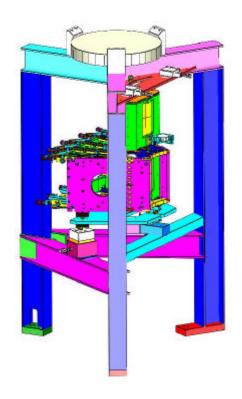
OAP2 is a single shell prototype with Titanium housings.
Alignment using CDA at GSFC



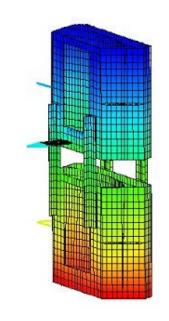
OAP2 FEA Model – in x-ray test orientation



OAP2 Vertical Assembly



OAP2 in vertical assembly facility

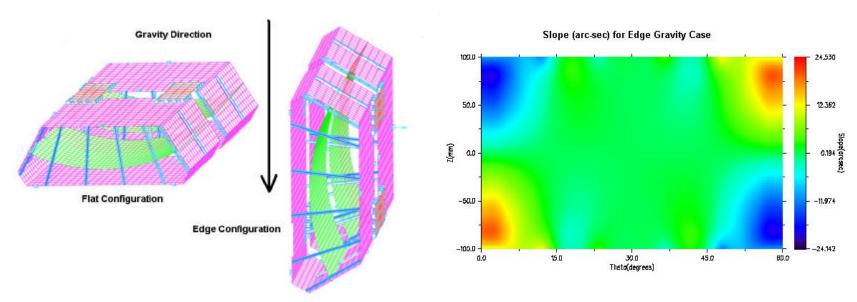


Gravity effects during vertical assembly add 1.5 arc-seconds (HPD) – in quadrature

OAP2 FEA Model – Assembly Orientation



Gravity Effects During OAP2 X-Ray Testing



OAP2 X-Ray Test Orientation:

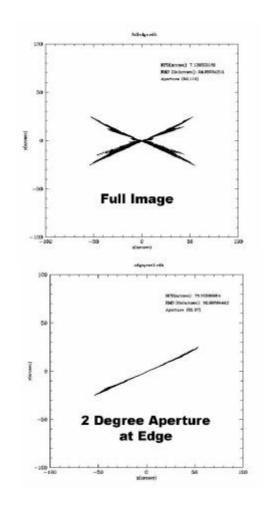
- Two orientations evaluated
- "Edge" Configuration selected based on performance

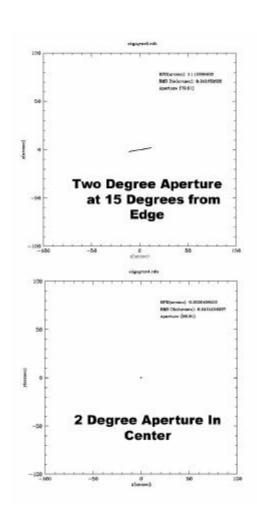
Load Case	HPD(50 Deg Aperture)				
1G Vertical, baseline support	1.5				
1G On edge (baseline)	7.0				
1G flat	17.6				

Gravity distortions are highly localized towards optic edges, allowing us to use a small aperture in the center to eliminate these effects



Gravity Distortion in X-ray Image





Aperture	HPD (arc-sec)
50 de: in center	7
2 deg: at edge	76
2 deg: 4 deg from edge	36
2 deg: 14 deg from edge	11
2 deg: in center	< 1



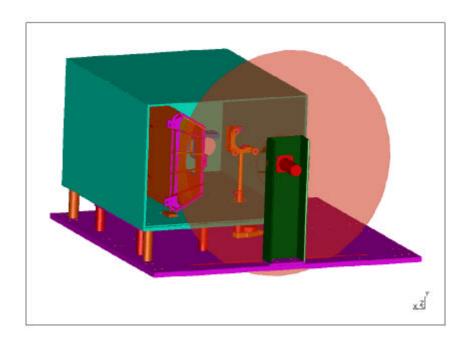
OAP2 Thermal Contributors

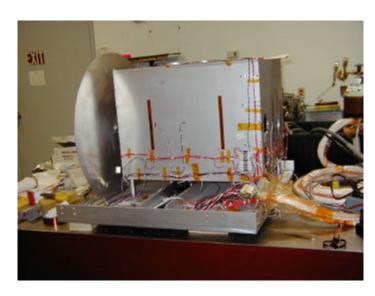
- OAP2 has glass optics with a CTE of 7.2 ppm/degrees C
- Housing material (Ti) chosen to minimize Glass/housing CTE mismatch (Ti is 9.5 ppm/degree C)
- OAP2 is sensitive to both differences in average temperature and to thermal gradients

	HPD Sensitivity	Tolerance	Contribution
Error Contributor	Arc-sec/ degree C	Degree C	HPD (Arc-seconds)
Thermal soak (per degree C)	0.3	1.0	0.3
Epoxy Bi-layer effects, per C	1.0	1.0	1.0
Axial Gradient (Z in Figure 5)	0.2	0.5	0.1
Transverse Gradient (Y in Figure 5) (Ti bars join P & H Hsgs)	7.3	0.1	0.73
Vertical Gradient (X in Figure 5)	1.5	0.5	0.75
RSS			1.48



OAP2 Thermal Control System





- Thermal control system and motor driven aperture wheel to be used in OAP2 x-ray testing.
- Evaluate thermal sensitivities by varying control set-points
- Evaluate gravity effects by scanning a small aperture over the extent of the OAP2 aperture